

1. (Currently Amended) A hybrid blade (1) for thermal turbomachines, having the blade comprising:
an airfoil (2)-made of a first metallic material of a certain first density, and having;
a blade root (3), characterized in that the blade root (3), compared with the airfoil (2), is made of a different second metallic material having a lower second density lower than the first density; and in that
wherein the airfoil (2) is connected to the blade root (3) in a positive-locking manner.
2. (Currently Amended) The hybrid blade (1) as claimed in claim 1, characterized in that said wherein the blade (1) is comprises a compressor blade, in particular a high-pressure compressor blade, in which:
wherein the airfoil (2) is made of a stainless CrNi steel; and
wherein the blade root (3) is made of a material selected from the group consisting of a high-temperature titanium alloy, or an intermetallic gamma titanium aluminide alloy, or and an intermetallic orthorhombic titanium aluminide alloy.
3. (Currently Amended) The hybrid blade (1) as claimed in claim 1, characterized in that said wherein the blade (1) is comprises a turbine blade; in which
wherein the airfoil (2) is made of a superalloy, for example a nickel-based superalloy; and
wherein the blade root (3) is made of a material selected from the group consisting of a high-temperature titanium alloy, or an intermetallic gamma titanium aluminide alloy, or and an intermetallic orthorhombic titanium aluminide alloy.
4. (Currently Amended) The hybrid blade (1) as claimed in one of claims 1 to 3claim 1, characterized in that said wherein the blade (1) is comprises a moving blade.
5. (New) The hybrid blade as claimed in claim 2, wherein the compressor blade

comprises a high-pressure compressor blade.

6. (New) The hybrid blade as claimed in claim 3, wherein the superalloy comprises a nickel-based superalloy.